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## Code Verification of ReFRESCO using the Method of Manufactured Solutions

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#### 1. Introduction

- Development of a reliable CFD solver requires thourough Code Verification to guarantee the correctness of the code and to assess its grid and time-step convergence properties
- Code Verification of a (U)RANS solver requires the use of the Method of the Manufactured Solutions to allow the evaluation of discretization errors



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#### 2. Manufactured Solutions

- Two-dimensional solutions that mimic near-wall, statistically-steady, incompressible flows
- Flow field includes a "linear sub-layer" for *y*<sup>+</sup><5 and the "skin friction coefficient" at the wall matches an empirical correlation for a flat plate turbulent boundary-layer
- Turbulence quantities of eddy-viscosity models are also manufactured (but not used in the present exercise)





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#### 3. ReFRESCO

- URANS solver with a fully-collocated arrangement and a face-based data structure
- Finite-volume discretization in the physical space
- Able to handle volumes of arbitrary shape, which means that it is suitable for complex geometries
- Preserving second-order grid convergence (at internal and boundary cells) is a challenge



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- 4. Non-orthogonality corrections
- Diffusion term of general transport equation

$$-\int_{S}\Gamma\left(\vec{\nabla}\phi\cdot\vec{n}\right)dS$$

Finite-volume discretization

$$-\int_{S} \Gamma\left(\vec{\nabla}\phi \cdot \vec{n}\right) dS \cong -\sum_{j=1}^{n_{faces}} \Gamma_{f_{j}}\left(\vec{\nabla}\phi \cdot \vec{n}\right)_{f_{j}} S_{f_{j}}$$



• Interpolation of 
$$\overline{
abla} \phi_{\!_{f_j}}$$

$$\left(\frac{\partial \phi}{\partial n}\right)_f = \vec{\nabla} \phi_f \cdot \vec{n}_f = \vec{\nabla} \phi_f \cdot \left(\vec{\Delta}_o + \vec{\Delta}_n\right)$$

$$\begin{pmatrix} \frac{\partial \phi}{\partial n} \end{pmatrix}_{f} = \left( \frac{\phi_{c2} - \phi_{c1}}{d_{o}} \right)^{n} + \left( \vec{\nabla} \phi_{f} \cdot \left( \vec{n}_{f} - \vec{\Delta}_{o} \right) \right)^{n-1} \\ \left( \frac{\phi_{c2} - \phi_{c1}}{d_{o}} \right) \cong \vec{\nabla} \phi_{f} \cdot \vec{\Delta}_{o}$$









#### 5. Results

Grids

Grid	Internal		Horizontal		Vertical	
$\operatorname{Set}$	Faces		Boundary Faces		Boundary Faces	
	$\delta^o_{max}$	$\delta^o_{avg}$	$\delta^o_{max}$	$\delta^o_{avg}$	$\delta^o_{max}$	$\delta^o_{avg}$
A1	29	20	29	20	0	0
A2	55	40	55	40	0	0
A3	70	56	70	<b>56</b>	0	0
B1	35	23	29	20	6	3
B2	63	45	55	40	8	4
B3	80	60	70	56	10	5



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#### 5. Results

- Determination of L<sub>2</sub> and L<sub>∞</sub> norms of the errors of mean flow quantities  $(u_x, u_y, C_p)$
- Observed order of grid convergence and error constants determined from data of six finest grids
- Second-order QUICK scheme in convection
- No excentricity issues for present grid sets



10<sup>-5</sup>

2

 $h_i/h_1^3$ 

 $10^{-6}$ 

p= 2.0 T3 p= 2.0 T4 p = 2.0

4

5 6 p = 2.0

 $T3_{bc}$ p= 2.0

T4<sub>b</sub> p=2.0

7

8



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#### 6. Conclusions

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- Non-orthogonality corrections are mandatory to obtain a consistent discretization scheme
- Non-orthogonality should also be corrected at boundary faces. However, the "boundary error" may be negligible
- Thorough Code Verification is essential for the credibility of any CFD code